

Seventh Framework Programme Marie Curie Action "International Research Staff Exchange Scheme"

Paton Welding Institute Kiev, Ukraine

# Protocol of composite material and its components test

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# 1. The material of the reinforcing element and the resin

As the reinforcing element is used straight roving (fig. 1) from fiberglass EC 10 1680H-V10(168) [2], as the filler - epoxy resin K $\square$ A-X $\square$  [1]. Here and further in the text of the Protocol, the roving – is the single strand of fiberglass, consisting of ~ 1200 filaments.

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№	Parameter	ЕС 10 1680H-У10(168) <sup>3)</sup>	ЕС 10 1680Н-У11(168)
1	Linear density, teks	1760	1748
2	Moisture content,%	0.02	0.02
3	Substances deleted during roasting,%	1.0	0.81
4	Unit breaking load, mN/teks <sup>2)</sup>	599.1	653.4

Table 1. The results of roving EC 10 1680H testings<sup>1)</sup>

Notes:

<sup>1)</sup> Tested on 28.04.2005 by " Объединение стеклопластик " Ltd according to [4, 5]. For winding was used the roving EC 10 1680H-У10(168) with the lubricant У10, the results of the roving in combination with lubricant У11 are given for reference;

<sup>2)</sup> 1teks = 1 g/1 km;

<sup>3)</sup> Designation of the roving according to [2]: E - general purpose glass; C - continuous thread; 10 - nominal diameter of elementary filament,  $\mu$ m; 1680 - nominal linear density, teks; H - for winding; V10 - lubricant; 168 - linear density of complex threads, teks.



Figure. 1. The roving used as the reinforcing element: a - roving, laid in the reel; b - cross section of roving; c - the end of the roving in free state.

According to [2] minimal unit breaking load of the roving: 396 mN/teks (40 g/teks).

Theoretical breaking load is calculated as the product of the linear density and the unit breaking load:  $y_{10}$ :  $1760 \times 599.1 = 1054416 \text{ mN} = 1054.416 \text{ N} = 107.5205 \text{ kgf}$ .  $y_{10}$ :  $1748 \times 653.4 = 1143891 \text{ mN} = 1143.891 \text{ N} = 116.6444 \text{ kgf}$ .

Table 2. Characteristics of epoxy resin КДА-ХИ according to the certificate<sup>1)</sup>

N⁰	Characteristic	According to [1]	Actually
1	Appearance	Homogeneous liquid of light yellow to light brow mechanical inclusions an	medium viscosity, from wn color without nd clots
2	Viscosity by viscometer VZ-246 (6 mm nozzle diameter) at $25 \pm 2$ °C, min	0.5-2.0	0.1
3	Gel time (sec) of 2 grams at temperature: $180 \pm 2 \ ^{\circ}C$ $160 \pm 2 \ ^{\circ}C$	50-105 80-120	94

Note:

<sup>1)</sup>Date of manufacture 7.04.2005, warranty period - 12 months.

Manufacturer's information (NPP «Synthesis», Donetsk) on the strength characteristics of the polymer matrix of the resin КДА-ХИ after the standard mode curing (100...180 °C during 2 hours):

-flexural strength - 90...110 MPa; -tensile strength - 75...80 MPa; -impact strength - 5...10 Kj/mm<sup>2</sup>; -elongation - 5.5...8%.

Additional information.

- after winding on a cylindrical pipe the roving cross-sectional dimensions typically are  $3 \times 0.3$  mm;

- after winding on tube with a diameter of 320 mm the density of filling was 0.7...0.72;

- the tension of the roving with force > 5 kg results in breaking of some filaments;

- filling density is defined as the ratio of the weight of glass it has after thermal removal of the resin at a temperature of 400...450 °C during 2...3 h, to the initial weight of the material;

- polymerization of composite repair after its installing is done at a temperature of 120...180°C. The same temperatures was used for polymerization of composite test specimens.

## 2. Preparation of the tensile test specimens of composite material

To determine the mechanical properties of the composite material specimens in form loop were manufactured. Specimen consisted of two branches, a and b (fig. 2, 3). The loop formed by winding the roving, soaked with epoxy, on special rig (fig. 2a, 2e). During the winding the roving was stretched by applied force. (table 3).



Figure. 2. Preparation of the tensile specimens: a - loops 1... 5, wrapped up in a snap; b - specimens of  $\Pi 1...\Pi 5$  before the test; c - sample  $\Pi 1$ ; d - model of used rig; e - winding loops on the rig.



Figure. 3. The geometrical dimensions of the tensile specimen: \* - dimensions are the same for all specimens; <> average size; (a), (b), (c), (d), (e), (f), (g), (h) - sections of specimen.

	1							
Specimen	Π1	П2	П3	П4	П5	Average		
<sup>1)</sup> Preliminary tension of row	ving, $N_1, kg$		2	4	3	3.5	0.1	2.5
<sup>2)</sup> number of turns of the	roving		80	80	80	80	75	79
	(a)		7.3	6.8	6.7	7.0	8.0	7 2
	(b)	7.1	6.8	6.5	7.0	8.6	1.4	
	(c)	4.9	5.1	5.3	4.9	5.2	5 1	
width of the section,	(d)	4.9	5.0	4.9	5.2	5.5	5.1	
mm		(e)	4.8	4.6	4.8	5.0	5.0	
	round place	(g)	4.9	4.7	4.7	4.9	5.0	10
		(f)	4.9	4.7	5	4.8	5.0	4.9
		(h)	4.9	4.6	4.7	5.0	5.5	
Fracture location			(g)	(f)	(e)	(e)	(g)	

Table 3. Characteristics of tensile specimens.

Notes: <sup>1)</sup> average value, amplitude variation during winding  $\pm$  1.5 kg; <sup>2)</sup> number of turns may vary  $\pm$  3.

# 3. The test results of the composite material



Figure. 4. Tensile test: a - testing machine Instron 8802; b - specimen gripping; c - extensometer Biaxial 2620-614; d, e, f, j, g, h - features of specimens fracture.



Figure. 5. Specimen  $\Pi 4$  after the destruction: a - cross section (1 – at the section c, 2 - at the section a, (see Fig. 3); b - longitudinal splitting of branche a (see Fig. 3).

Table 4.	Results	of	tensile	tests <sup>1</sup>	.)
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Specimen	П1	П2	П3	$\Pi 4^{3)}$	П5	Average
stress of fracture, kN	95.28	101.23	99.19	96.04	99.87	98.32
max. strain,%	1.36	1.44	1.59	1.39	1.58	1.47
Poisson's ratio µ	0.179	0.196	0.164	-	0,147	0.172
<sup>2)</sup> Young's modulus, E, MPa	32721	34645	31673	33079	25518	31527
<sup>2)</sup> tensile strength, MPa	445.0	498.9	503.6	459.8	403.2	462.1

Notes:

<sup>1)</sup>Longitudinal and transverse base of was - 25 mm and 15 mm, respectively. Test temperature + 20 <sup>o</sup>C, speed of deformation 10 mm/min.

<sup>2)</sup> The thickness in the middle part of the loop was used in the calculation of the mechanical properties of composite; <sup>3)</sup> During the test, the shift of the extensometer had occurred.

In Figure 6 are given examples of the specimen  $\Pi 2$  tension.

From fig. 6d we can see that stress  $F_1$ , per one roving, almost in proportion to the longitudinal deformation, so after removing of the extensometer the destruction deformation can be found by linear continuation of relation up to ultimate load on one roving.

Summary of the test results of specimens  $\Pi 1...\Pi 5$ , given to the force on 1 roving, are presented in Figure 7.

It should be noted that dependence of the destruction and ultimate deformation on preloading of the roving was not detected (fig. 8).



Figure. 6. Tension of the specimen  $\Pi 2$ : a, b, d - dependence of force per 1 roving (F<sub>1</sub>) from time (t), displacement of grips ( $\Delta l$ ) and strain ( $\overline{e}$ ), respectively; c, e - transverse strain ( $\overline{e}^*$ ) and ratio ( $\overline{e}/\overline{e}^*$ ) depending on the longitudinal deformation ( $\overline{e}$ ). 1, 2 - test results; 3 - load-shedding; 4 - removal of extensioneter; 5 - ultimate load; 6 - linear relationship.

Table 5. Longitudinal deformation  $\overline{e}$  depending on the force per 1 roving F<sub>1</sub> and on the parameter  $E_1 = \frac{F_1}{\overline{a}}$ .

E								
	L	oad sheddii	ng	Remov	al of extens	someter	Fracture	
Specimen	$\overline{e}$	$F_1, N$	$E_1, N$	$\overline{e}$	$F_1, N$	$E_1, N$	$\overline{e}_{\max}$ calc.	$F_{1\max}, N$
П1	0.008453	380.8705	45057.44	0.010409	465.1008	44681.88	0.013328	595.5324
П2	0.008336	378.9419	45458.48	0.010417	465.8148	44714.94	0.01415	632.7281
П3	0.00835	357.5832	42824.34	0.010496	450.6579	42937.59	0.014439	619.9876
П4	0.008488	427.4154	50355.25	0.011518	556.4347	48308.12	0.012426	600.2789
П5	0.010759	449.5257	41781.36	0.015899	665.8315	41877.96	0.015899	665.8315
Average	0.008877	398.8673	45095.37	0.011748	520.7679	44504.1	0.014049	622.8717

Notes:  $F_{1\text{max}}$  - fracture force per 1 roving;  $\overline{e}_{\text{max}}$  calc. - design value of ultimate deformation; the value of  $E_1$  is the same as at the moment of the extension removing.

Table 6. Some clarifications of the tensile test.

Specimen	Fracture force, N	Ratio $\left  \overline{e}_{*} / \overline{e} \right $ For $\overline{e} >>0$ (see fig. 6e)	The tangent of an angle α (see fig. 6c)
П1	95285.18	0.18	0.18
П2	101236.5	0.19	0.17
П3	99198.01	0.17	0.17
П4	96044.63	-	-
П5	99874.73	0.15	0.15
average	98327.81	0.173	0.168



Figure 7. Tension of 1 roving (samples of  $\Pi 1...\Pi 5$ ): 1 - according to extensioneter; 2 - load shedding; 3 - removal of extensioneter; 4 - linear continuation from the condition of proportionality; 5 - limit values the force  $F_1$  and strain  $\overline{e}$ ; 6 - the average maximum value; 7 - average dependence of force  $F_1$  from strain.



Figure 8. The dependence of the force  $F_{1max}$  (applied to 1 roving) - (a) and strain  $\overline{e}_{max} calc$  - (b) from the tension of the roving N<sub>1</sub> (see Table 3);  $F_{1max}$  from  $\overline{e}_{max} calc$  - (c).

1 - fracture force of the preloaded roving; 2 - ultimate strain of preloaded roving; 3 - the force and strain at fracture; 4 - average maximum values.

# 4. The tensile test of the roving

In addition to the composite test the roving test was conducted (Fig. 9). The test was carried out on machine Instron 8802 using extensioneter 2620-601. Measuring base was 52.5 mm, working section length - 140 mm. Temperature during the test + 20  $^{\circ}$ C, speed of displacement - 30 mm/min.



Figure 9. Tensile test of roving: a - testing machine Instron 8802; b - extensioneter 2620-601 installed on the specimen; c - general view of the roving test; d - the "snail" grip.

As a specimen was used a section of the roving spooled from reel.

Limit values of strains (Table 7) were obtained by calculation and not by direct measurements. Mechanical properties of the roving were determined on the base of test results of specimens P1, P2, P3 (Table 7). Test results of specimens are presented in Figure 11.

The graphical dependency  $\overline{\overline{e}} = f(\Delta l)$  was not available because of instability of extensioneter fixation.



Figure 10. Specimens of roving (P1, P2, P3) after tensile testing.

# specimen	fracture force, kN	Max. deformation, %	Rigidity, kN/mm <sup>1)</sup>	Specimen marking	Place of destruction
$1^{2)}$	0.41	1.73	0.91		in grips
2	0.62	1.64	1.55	P1	working part
3	0.64	1.20	0.88		in grips and working part
4	0.55	1.02	1.04		in grips and working part
5 <sup>3)</sup>	0.66	-	-		working part
6 <sup>3)</sup>	0.73	-	-		working part
7 <sup>3)</sup>	0.65	-	-		working part
8	0.69	1.31	1.21	P2	working part
9	0.6	1.70	_	P3	working part

Table 7. The results of the roving tensile test.

Note: <sup>1)</sup>Approximate value; <sup>2)</sup>The sample had the original damage; <sup>3)</sup>Tests without the use of extensometer.



Figure. 11. Dependence of force in specimens P1, P2, P3 from grips displacement  $\Delta l$ .  $F_{1 \text{ max}}$  - maximal force during the test; Lin. - linear dependence.



#### 5. Comparison of test results of the roving and composite material

Figure. 12. A summary diagram of tension. Composite test: 1 - limit force  $F_1$  and strain  $\overline{e}$ ; 2 - average maximum value; 3 - average dependence  $F_1 = f(\overline{e})$ . Roving test (specimens P1, P2, P3): 4 - displacement by extensometer; 5 - load-shedding; 6 - extensometer taking-off; all specimens in Table 7 except specimen 1; 8 - average of all data in Table 7.

Table 8. Test results of the roving specimens and roving as component of  $\Pi 1...\Pi 5$  specimens.

Roving					Roving of II1II5 specimens			
Specimen	$\overline{e}$	$F_1, H$	$E_1, H$		Specimen	$\overline{e}$	$F_1, H$	$E_1, H$
P1	0.0164	620	37805		П1	0.0133	595.5	44682
P2	0.0131	690	52672		П2	0.0142	632.7	44715
P3	0.0170	600	35294		П3	0.0144	620.0	42938
					П4	0.0124	600.3	48308
					П5	0.0159	665.8	41878
Average	0.0155	636.7	41924		Average	0.0140	622.9	44504
					min	0.0124	595.5	41878
					max	0.0159	665.8	48308

In calculations could be used average values of specimens  $\Pi 1...\Pi 5$  test data: the limit strain  $\overline{e} = 0,014$ ; the limit force per roving  $F_1 = 622,9 H$ , and from here,  $E_1 = F_1/\overline{e} = 622,9/0,014 = 44493 H$ .

## 6. Additional research of the material

## Linear density of the roving

A sample of roving with length of 1 m was weighed on Denver Instrument APX-60 (maximum permissible weight 60 g, measuring accuracy 0.1 mg.) Error in finding the length of specimen:  $\pm 1$  mm. Results are listed in table 9.

Table 9. Weight of 1 m roving.

<u>№</u> weighing	1	2	3	4	5	6	7	8	9	Average
Weight, gr.	1.7214	1.7215	1.7212	1.7214	1.7215	1.7214	1.7214	1.7214	1.7215	1.7214



Figure 13. Specimen of the roving with length 1 m, subjected to weighting.

Linear density of roving, according to measurements - 1721.4 teks (according to the certificate (see table 1) - 1760 teks).

#### Measurement of the cross section of the roving in a free state

The measurement was carried out using a caliper. The following results were obtained: width ~ 5 mm, thickness ~ 0.3 mm. Diameter of twisted roving ~ 1 mm.

#### Photomicrography of the roving

The pictures were used to determine the diameter of the roving filaments. Based on the results of the measurements filament diameter is  $\sim 10 \mu m$ . It was noted the transparency of the filaments (Fig. 14).



Figure. 14. Roving filaments (enlarged).

#### Thermal testing of the composite

After the destruction the elements of samples  $\Pi 1..\Pi 5$  were subjected to half-hour exposure to different temperatures (Fig. 15). At temperature 250 °C composite slightly darkens, at 300 °C it darkens and begins to give off unpleasant smell, at 350 °C it emits strong smell and starts spilling out.



Figure. 15. Elements of the specimens after half an hour exposure excerpts to different temperatures.

## Measurement of the cross-sectional area of roving

At dense packing of the roving with tension ~ 5 kg in the groove (cross section  $19.2 \times 3.1 \text{ mm}$ ) total number of strands of the roving amounted to 57 (fig. 16). Coming from here, the cross-sectional area of the roving in wrapped state -  $\frac{19,2\times3,1}{57}$ =1,044 mm<sup>2</sup>.

Assuming that the cross section of the filament is 6-sided polygon of 10 microns diameter we would obtain cross-sectional area of the roving (1200 filaments)  $0.416 \text{ mm}^2$  - most dense packing. In this case the groove could be filled up by 143 strands. The discrepancy between the practical and the theoretical density of packing is most likely be explained by real form of cross section of the

filaments (fig. 14b), the presence of lubricant and insufficient stress tension during laying. For the complete straightening of the filaments of the roving applied tension should be 150...200 N (Fig. 11).

Note that average area per 1 roving of the specimens was at sections: (a), (b) -  $1.36 \text{ mm}^2$ ; (c), (d) -  $0.96 \text{ mm}^2$ ; (e), (f), (g), (h) -  $0.92 \text{ mm}^2$  (Fig. 3).



Figure. 16. Laying of roving in the groove: a - groove cross-section  $19.2 \times 3.1$  mm; b - roving, laid into the groove.

## 7. Properties of the filler material

According to the data provided by MPRI NASB in the calculations, could be used the following (approximate) characteristics of filler:

-Young's modulus: 1.8-2.2 HPa; -yield strength: 80 MPa; -Poison's ratio: 0.25 -impact strength 15 kJ/m<sup>2</sup>. -density: 1000-1100 kg/m<sup>3</sup>.

## 8. List of references

1. TU UA 24.6-0030314547-002-2004- Ероху resin КДА - ХИ. (ТУ У 24.6-0030314547-002-2004 - Связующее эпоксидное марки КДА - ХИ).

2. GOST 17139-2000-Fiberglass. Roving. Specification. Minsk. (ГОСТ 17139-2000 - Стекловолокно Ровинги Технические условия. Минск).

3. Gutnikov S. I., Lazorak B. I., Seleznyov A. N. Glass fibers: Handbook for students studying in composite nanomaterials. -М., 2010. -53 s. (Гутников С. И., Лазоряк Б. И., Селезнев А. Н. Стеклянные волокна : учеб. пособие для студентов по специальности «Композиционные наноматериалы». - М., 2010. - 53 с).

4. GOST 6943.10-79 - Textile glass. Method for determination of breaking load and elongation at break. (ГОСТ 6943.10-79 - Материалы текстильные стеклянные. Метод определения разрывной нагрузки и удлинения при разрыве.).